



Internal Nuclear Propulsion System for Future Space Applications

Ramkumar K S*

Chandigarh University, Chandigarh State, Punjab-140413, India ORCID: 0009-0007-0153-971X

Samandeep Kourt

Chandigarh University, Chandigarh State, Punjab-140413, India ORCID: 0009-0000-0576-0138

Abstract: Internal nuclear pulse propulsion (INPP) represents a novel approach to space propulsion, where controlled nuclear detonations occur within the spacecraft itself, offering potential advantages in safety and propulsion control compared to conventional external detonation designs like Project Orion. INPP relies on the same fundamental principles of nuclear pulse propulsion, utilizing the momentum generated by nuclear explosions to propel the spacecraft forward. Proposed INPP designs involve strategically placing small nuclear devices within the spacecraft's structure, detonating them in sequence to produce controlled impulses for propulsion. However, significant engineering challenges, such as radiation shielding, precise detonation control, and safety considerations, must be addressed to realize the full potential of INPP. Despite these hurdles, ongoing research and development efforts aim to harness the transformative capabilities of INPP, promising to revolutionize space exploration by enabling efficient and rapid interstellar travel while ensuring the safety of spacecraft and crew.

Table of Contents

1. Introduction	1
2. Working Principle	2
3. Fuel Selection.	2
4. Reactor and Core Geometry	3
5. Cooling and Radiation Shielding	4
6. Thrust Equation	5
7. Conclusion	5
8. Future Scope of this Work	6
9. References.	6
10. Conflict of Interest and Funding	6
11. Funding Information	6

1. Introduction

In the pursuit of exploring the vast expanse of space and reaching distant celestial bodies, the development of advanced propulsion technologies has remained a cornerstone of scientific inquiry and innovation. Among these technologies, Internal Nuclear Pulse Propulsion (INPP) stands out as a promising concept with the potential to revolutionize space exploration by enabling spacecraft to achieve unprecedented speeds and traverse vast interstellar distances. INPP represents a departure from conventional propulsion methods by harnessing the immense energy released from controlled nuclear reactions within the propulsion system itself. Unlike external nuclear propulsion systems like Project Orion, which utilize nuclear explosions outside the spacecraft to generate thrust, INPP operates through the precise management of nuclear reactions within the core of the rocket. The fundamental principle behind INPP involves the utilization of nuclear energy to propel a spacecraft forward. This is achieved through a series of controlled nuclear reactions, such as nuclear fission or fusion, which release energy in the form of high-velocity radiation. By directing and harnessing this radiation through clever engineering and design, INPP systems can generate thrust and propel spacecraft at velocities that far exceed those achievable with traditional chemical propulsion systems.

^{*}PG Research Scholar, Chandigarh University, Chandigarh State, Punjab-140413, India. **Corresponding Author: ksramkumar276@gmail.com**. *PG Research Scholar, Chandigarh University, Chandigarh State, Punjab-140413, India. **Contact: kaursamandeepk@gmail.com**

^{**}Received: 15-May-2024 || Revised: 28-May-2024 || Accepted: 29-May-2024 || Published Online: 30-May-2024.

The concept of INPP has captured the imagination of scientists, engineers, and space enthusiasts alike due to its potential to enable rapid interplanetary and interstellar travel. With the ability to achieve higher speeds and shorter travel times, INPP offers the tantalizing prospect of exploring distant worlds within our own solar system and venturing beyond to explore exoplanets and distant star systems. However, the realization of INPP as a viable propulsion technology presents numerous technical, engineering, and logistical challenges. From designing robust and efficient propulsion systems to ensuring the safety and reliability of nuclear reactors in space, the development of INPP requires interdisciplinary collaboration and rigorous scientific inquiry. This research paper seeks to provide a comprehensive overview of Internal Nuclear Pulse Propulsion, exploring its theoretical foundations, technical feasibility, potential applications, and the challenges and opportunities associated with its development. By delving into the intricacies of INPP, we aim to shed light on this cutting-edge propulsion concept and its role in shaping the future of space exploration.

2. Working Principle

The principle of conservation of momentum applies to Internal Nuclear Pulse Propulsion (INPP) systems when considering the expulsion of gamma radiation to generate thrust. According to this principle, the total momentum of a closed system remains constant unless acted upon by external forces. In the context of INPP, the momentum of the expelled gamma radiation contributes to the propulsion of the spacecraft. When nuclear reactions occur within the propulsion system's core, they release energy in the form of gamma radiation, which consists of high-energy photons. These gamma rays have momentum due to their energy and speed. As they are emitted from the spacecraft, they carry momentum in a specific direction, exerting a force on the spacecraft in the opposite direction according to Newton's third law of motion.

For example, during nuclear decay processes or interactions within the reactor core, gamma rays are emitted. By expelling these gamma rays at high velocities, the spacecraft experiences an equal and opposite force, propelling it forward. The conservation of momentum by gamma radiation in INPP systems is crucial for understanding how thrust is generated and controlled. By carefully managing the direction, energy, and emission rate of gamma radiation, engineers can optimize thrust efficiency and maneuverability for space missions. In summary, the principle of conservation of momentum by gamma radiation dictates that the momentum imparted to the expelled photons contributes to the propulsion of the spacecraft in INPP systems. This principle underscores the fundamental mechanism by which INPP achieves thrust and highlights the importance of balancing momentum conservation principles with the design and operation of propulsion systems for space exploration.

3. Fuel Selection

In Internal Nuclear Pulse Propulsion (INPP) systems, the fuel plays a critical role in enabling controlled nuclear reactions to generate thrust and propel the spacecraft forward. Several key requirements must be met for the fuel to be suitable for use in INPP: Firstly, the fuel must contain fissile isotopes capable of sustaining nuclear fission reactions. Fissile materials, such as uranium-235 (U-235) or plutonium-239 (Pu-239), undergo fission when bombarded with neutrons, releasing energy in the process. This energy release serves as the primary source of propulsion in INPP systems. Secondly, the fuel should possess a high energy density to maximize the amount of energy that can be extracted per unit mass or volume. High energy density fuels allow for compact reactor designs and increased propulsion efficiency, enabling longer missions and higher velocities.

Additionally, the fuel must exhibit stability and reliability under the extreme conditions encountered during space missions. This includes withstanding high temperatures, radiation exposure, and mechanical stresses without undergoing unwanted reactions or degradation. Moreover, the fuel should be readily available and accessible for use in INPP systems. Accessibility considerations include ease of handling, transport, and storage of the fuel both on Earth and in space. Furthermore, safety is paramount in INPP systems, and the fuel should pose minimal risk of accidental release, contamination, or radiation exposure. This includes selecting fuels with low probabilities of criticality accidents or radioactive leakage. Additionally, the fuel must be compatible with other components of the propulsion system, including reactor materials, coolant systems, and thrust mechanisms. Compatibility considerations ensure efficient operation and minimize the risk of corrosion or other adverse effects. Lastly, the fuel selection must comply with regulatory requirements and international agreements governing the use of nuclear materials in space exploration. This includes adherence to safety standards, environmental regulations, and non-proliferation agreements. By meeting these requirements, the fuel can effectively support the

operation of INPP systems, providing the necessary energy for propulsion while ensuring safety, reliability, and compliance with regulatory standards.

Selecting uranium as a fuel in Internal Nuclear Pulse Propulsion (INPP) offers several advantages due to its unique properties and availability:

High Energy Density: Uranium is a highly dense material, capable of storing a large amount of energy within a relatively small volume. The energy density of uranium fuel allows for efficient propulsion systems that can achieve high speeds and cover large distances in space missions.

Abundance: Uranium is relatively abundant in the Earth's crust, providing a stable and consistent supply of fuel for INPP systems. This abundance ensures reliable access to fuel sources, reducing the dependency on scarce or expensive resources.

Fissile Capability: Uranium contains fissile isotopes, such as uranium-235 (U-235), which can sustain nuclear fission reactions. Through controlled fission processes, U-235 releases a significant amount of energy, which serves as the basis for propulsion in INPP systems.

Conversion to Plutonium-239: In breeder reactors, U-235 can be converted into plutonium-239 (Pu-239) through neutron capture and subsequent beta decay processes. Pu-239 is also fissile and can undergo nuclear fission reactions, providing additional fuel options for INPP systems and extending the availability of nuclear fuel resources.

Safety and Reliability: Uranium-based fuels offer well-understood safety characteristics and proven reliability in nuclear applications. The safety record of uranium fuels, combined with established operational practices and regulations, ensures the safe and effective operation of INPP systems.

In summary, selecting uranium as a fuel in INPP offers a combination of high energy density, abundance, and fissile capability, conversion potential to Pu-239 in breeder reactors, and safety and reliability. These factors make uranium a practical and efficient choice for powering spacecraft propulsion systems in future space exploration missions.

4. Reactor and Core Geometry

In Internal Nuclear Pulse Propulsion (INPP), pulsed fission propulsion systems hold considerable promise for propelling spacecraft efficiently through space. These systems utilize controlled nuclear fission reactions within the core to generate thrust. The core geometry commonly considered for such systems is cylindrical. For a controlled reaction, control rods, composed of neutron-absorbing materials like boron or cadmium, are pivotal for regulating nuclear reactions in INPP reactors. Their insertion into the reactor core modulates neutron flux, enabling precise control over reactor power levels. In emergencies, these rods can be fully inserted to swiftly halt the chain reaction, ensuring safety. Additionally, INPP reactors feature emergency shutdown mechanisms, including scram rods and Reactor Protection Systems (RPS). Scram rods, actuated hydraulically or pneumatically, rapidly insert into the core to terminate reactions. RPS, equipped with sensors and automated algorithms, monitor reactor parameters and trigger shutdowns if abnormal conditions arise. These redundant safety systems uphold reactor integrity, protecting spacecraft and crew from radiation hazards, and ensuring mission success.

Pulsed Fission Propulsion: Pulsed fission propulsion harnesses the energy released from nuclear fission reactions to propel spacecraft forward. Nuclear fission generates high-velocity particles, such as neutrons and gamma rays, which impart momentum to the spacecraft, resulting in thrust. These systems offer several advantages:

High Energy Output: Nuclear fission reactions release substantial energy, providing high thrust levels suitable for efficient propulsion.

Controllable Thrust: Pulsed fission propulsion allows precise control over thrust pulses, facilitating maneuverability during missions.

Efficient Energy Conversion: These systems efficiently convert nuclear energy into kinetic energy, maximizing propulsion efficiency.

Safety: Pulsed fission propulsion minimizes radiation exposure risks by operating in a pulsed mode.

Feasibility: Extensive research has demonstrated the feasibility and potential for practical implementation of pulsed fission propulsion in space missions.

Cylindrical Core Geometry

The cylindrical core geometry is favored for INPP systems utilizing pulsed fission propulsion. In this configuration, nuclear fuel elements are arranged cylindrically within the core, offering numerous benefits:

Uniform Energy Distribution: Cylindrical cores ensure even distribution of nuclear fuel elements, optimizing energy utilization and thermal management.

Compactness: Cylindrical designs allow for compact reactor layouts, maximizing space utilization while maintaining high thrust output.

Scalability: These designs are scalable to accommodate various spacecraft sizes and mission requirements, providing versatility.

Structural Integrity: Cylindrical configurations offer stability and structural integrity, capable of withstanding propulsion forces.

Ease of Manufacturing: Cylindrical core geometries are straightforward to manufacture, reducing complexity and cost.

5. Cooling and Radiation Shielding

In Internal Nuclear Pulse Propulsion (INPP), a regenerative cooling system is essential for managing the high temperatures generated by nuclear reactions within the propulsion system's core. This cooling system circulates a coolant, such as liquid hydrogen or liquid metal, through channels within the reactor core to absorb heat and prevent the overheating of critical components. The heated coolant is then directed to a heat exchanger, where it transfers its thermal energy to the propellant before being recirculated back into the core. This closed-loop cooling process ensures efficient thermal management and prolongs the operational lifespan of the propulsion system.

Benefits of Regenerative Cooling System:

- *Heat Dissipation:* Regenerative cooling effectively dissipates heat generated by nuclear reactions, preventing excessive temperatures that could damage reactor components.
- *Thermal Efficiency:* By transferring heat from the reactor core to the propellant, regenerative cooling enhances the thermal efficiency of the propulsion system, maximizing thrust output per unit of fuel consumed.
- *System Reliability:* The closed-loop design of regenerative cooling minimizes the risk of coolant leakage and system failure, ensuring reliable operation during space missions.
- *Longevity:* Efficient thermal management provided by regenerative cooling extends the operational lifespan of the propulsion system, reducing maintenance requirements and enhancing mission duration.
- *Adaptability:* Regenerative cooling systems can be tailored to accommodate different coolant types and reactor configurations, offering flexibility in design and implementation for various mission profiles.

Radiation Shielding Material: Lead (Pb)

Lead (Pb) is a commonly used radiation shielding material due to its high density and effectiveness in attenuating gamma rays and X-rays. Its ability to absorb and scatter ionizing radiation makes it an ideal choice for protecting spacecraft components and crew members from the harmful effects of radiation exposure in space. Lead-based shielding materials offer the following advantages:

- *High Density:* Lead possesses a high atomic number (82), resulting in dense shielding that efficiently attenuates ionizing radiation.
- *Gamma Ray Attenuation:* Lead effectively attenuates gamma rays through the processes of photoelectric absorption, Compton scattering, and pair production, reducing radiation exposure to acceptable levels.
- *Versatility:* Lead shielding can be easily molded or fabricated into various shapes and configurations to fit the specific requirements of spacecraft components and crew habitats.
- *Cost-Effectiveness:* Lead is readily available and cost-effective compared to other radiation shielding materials, making it a practical choice for space missions with budget constraints.

• **Durability:** Lead-based shielding materials are durable and resistant to degradation, providing long-term protection against radiation hazards throughout the duration of space missions.

6. Thrust Equation

To derive the thrust equation for Internal Nuclear Pulse Propulsion (INPP), we can start with the basic principles of momentum conservation. INPP involves the expulsion of radiation particles, such as photons or charged particles, to generate thrust. According to Newton's third law of motion, for every action, there is an equal and opposite reaction. Therefore, the momentum imparted to the expelled particles contributes to the propulsion of the spacecraft.

Let's denote:

- > m_p as the mass of the expelled particles per pulse,
- \triangleright v_e as the exhaust velocity of the particles,
- ▶ N as the number of pulses per unit time, and
- \succ F as the thrust generated.

Using the definition of thrust F as the rate of change of momentum, we have:

$$F = \frac{dp}{dt}$$

where p is the momentum. For INPP, the momentum change dp per pulse is equal to the momentum imparted to the expelled particles, which can be expressed as:

$$dp = m_p * v_e$$

Now, if we consider N pulses per unit time, the total thrust F generated is the product of the momentum change per pulse and the number of pulses per unit time:

$$F = N * m_p * v_e$$

This equation gives the thrust generated by INPP, taking into account the mass of the expelled particles per pulse m_p , their exhaust velocity v_e , and the frequency of pulses N.

In practice, the thrust equation can be further refined by considering factors such as the efficiency of the propulsion system, the specific characteristics of the nuclear reactions involved, and the design parameters of the spacecraft. However, this simplified derivation provides a basic understanding of how thrust is generated in INPP systems through the expulsion of radiation particles.

The above equation is a general equation where the momentum of particles, the mass of expelled particles, and velocity need to be calculated experimentally. The value of N should be decided according to the rate of reaction allowed for a stable reactor with their control systems.

7. Conclusion

In conclusion, Internal Nuclear Pulse Propulsion (INPP) represents a promising avenue for advancing space exploration capabilities and unlocking the potential for ambitious interplanetary and interstellar missions. Throughout this paper, we have explored the principles, design considerations, and key components of INPP systems, highlighting their potential to revolutionize space propulsion technology. INPP harnesses controlled nuclear reactions to generate thrust, offering high efficiency and energy density compared to traditional chemical propulsion systems. By expelling radiation particles at high velocities, INPP systems can achieve rapid acceleration and enable shorter travel times to distant destinations within the solar system and beyond.

The selection of appropriate reactor designs, such as pulsed fission propulsion with cylindrical core geometries, along with advanced cooling and radiation shielding technologies, is crucial for ensuring the safety, reliability, and efficiency of INPP systems. Control mechanisms, including control rods and emergency shutdown systems, play pivotal roles in regulating nuclear reactions and maintaining stable operation. Looking ahead, the future of INPP holds immense potential for advancing space exploration capabilities. Continued research and development efforts will focus on optimizing reactor designs, enhancing propulsion efficiency, and addressing safety and environmental considerations. Advancements in materials science, reactor technologies, and propulsion systems

will pave the way for the realization of ambitious missions to explore distant planets, asteroids, and even neighboring star systems. Moreover, the integration of INPP with other emerging technologies, such as artificial intelligence, robotics, and additive manufacturing, will further enhance mission capabilities and enable autonomous and adaptive spacecraft operations. Collaboration between government agencies, research institutions, and private aerospace companies will be essential for driving innovation and overcoming technical challenges associated with INPP. In conclusion, INPP holds the promise of unlocking new frontiers in space exploration, enabling humanity to venture further into the cosmos and expand our understanding of the universe. With continued research, investment, and collaboration, INPP has the potential to revolutionize space travel and shape the future of human civilization beyond Earth's boundaries.

8. Future Scope of this Work

The future of INPP is indeed brimming with exciting possibilities and potential avenues for exploration and innovation. Here are some key future scopes for INPP:

Interplanetary Missions: INPP systems could enable faster and more efficient travel to destinations within the solar system, including manned missions to Mars, robotic missions to the outer planets, and asteroid exploration missions.

Interstellar Exploration: Further advancements in propulsion technology could pave the way for interstellar missions to neighboring star systems, potentially enabling humanity to explore exoplanets and search for signs of extraterrestrial life.

Space Resource Utilization: INPP systems could facilitate the extraction and utilization of resources from celestial bodies, such as water ice from asteroids or the Moon, to support long-duration space missions and establish sustainable habitats beyond Earth.

Space Infrastructure Development: Utilizing INPP systems could aid in the construction of large-scale space infrastructure, such as space stations, habitats, and solar power satellites, opening up new opportunities for space-based industries and commercial activities.

Environmental Monitoring and Planetary Defense: INPP-powered spacecraft could be employed for environmental monitoring, planetary defense, and space situational awareness, helping to safeguard Earth from potential asteroid impacts and mitigate the effects of climate change.

Scientific Exploration: INPP missions could enable groundbreaking scientific discoveries in fields such as astrophysics, planetary science, and astrobiology, shedding light on the origins and evolution of the universe and the potential for life beyond Earth. Overall, the future of INPP holds immense potential for advancing humanity's presence in space, expanding our scientific knowledge, and fostering international cooperation in the exploration and utilization of outer space resources. With continued research, investment, and collaboration, INPP has the power to shape the course of human space exploration for generations to come.

9. References

- Schmidt, G. R., Bonometti, J. A., & Irvine, C. A. (2002, May). Project Orion and Future Prospects for Nuclear Pulse Propulsion. Journal of Propulsion and Power, 18(3), 497–504.
- [2] Nance, J. C. (1965). Nuclear Pulse Propulsion. IEEE Transactions on Nuclear Science, 12(1), 177–182.
- [3] Winterberg, F. (2009, June). Advanced charged particle beam ignited nuclear pulse propulsion. Acta Astronautica, 64(11–12), 1080–1084.
- [4] Grey, J. (1966, December). NUCLEAR-PULSE PROPULSION—I. Annals of the New York Academy of Sciences, 140(1), 393–395.
- [5] Sforzini, R. H. (1970, November). Derivation of the thrust equation from conservation of energy. Journal of Aircraft, 7(6), 538–541.

10. Conflict of Interest and Funding

The author declare no competing conflict of interest.

11. Funding Information

No external funding was received to support this study.